# A shear deformation theory of beams of bisymmetrical cross sections based on the Zhuravsky shear stress formula

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#### ABSTRACT

The paper is devoted to simply supported beams with bisymmetrical cross sections under a generalized load. Based on the Zhuravsky shear stress formula the shear deformation theory of a planar beam cross section is formulated. The deflections and the shear stresses of example beams are determined. Moreover, the numerical-FEM computations of these beams are carried out. The results of the research are shown in figures and tables.

Key words: Shear deformation theory; Shear stress; Beam deflection; Shear effect

## 1. Introduction

The problem of shear stress in bending of beams was recognized and described by D.I. Zhuravsky in the 19th century. GERE and TIMOSHENKO [1] described in detail the Zhuravsky's shear stress formula. S.P. Timoshenko in 1921 initiated an important problem consisting in considering the shearing effect in bending of beams by generalized Euler-Bernoulli beam theory. He assumed that the normal straight line before bending remains a straight line after bending being no more perpendicular to the bent beam line. The theory of normal straight line deformation of bent beams and plates was intensively developed in the latter half of the 20th century. RYCHTER [2] presented the shear-deformation beam-bending theories and extended them to two-dimensional displacement and stress fields. The relationships between accuracy, shear factors, and physical interpretations of the theory have been determined, showing that the Bernoulli-Euler type theory is a special case, giving accuracy equal to that of the other shear-deformation theories. KATHNELSON [3] considered small static deformation of isotropic linear elastic beams subjected to bending under arbitrary transverse load varying along the axial coordinate. A three-dimensional variational equation was used to formulate the governing equations of Timoshenko type, allowing to derive the formulae for the displacements and stresses in the cross-sectional area. WANG et al. [4] presented the theories of normal straight line deformation taking into account the shearing effect for beams and plates, with consideration of the papers published in past decades. HUTCHINSON [5] derived a new formula for the shear coefficient. In case of a circular cross section, the resulting shear coefficient is perfectly compatible with the values obtained by most of the publications. The shear coefficients were found for a number of various beam cross sections. KADOLI et al. [6] studied static behavior of functionally graded metal-ceramic beams using the higher order shear deformation theory. The principle of stationary potential energy was used in order to derive the static equilibrium equation of the beam in the finite element form. It was shown that the beam behaviour depends on whether the load is applied to the ceramic rich face or metal rich face of the beam. JUN and HONGXING [7] dealt with a uniform laminated composite beam using the trigonometric shear deformation theory, providing for sinusoidal variation of the axial displacement over the cross-section. Solution of the governing differential equations of motion enabled to formulate the dynamic stiffness matrix. REDDY [8] used the differential constitutive relations of Eringen and the von Kármán nonlinear strains in order to improve the classical and shear deformation beam and plate theories. This allowed to develop a new finite

element model. Such an approach enabled to determine the effect of geometric nonlinearity and nonlocal constitutive relations on bending response of these structures. GHUGAL and SHARMA [9] analyzed static bending of thick isotropic beams with consideration of the transverse shear deformation effects. The authors applied the hyperbolic shear deformation theory in order to avoid the use of the shear correction factor. The principle of virtual work allowed to formulate the governing differential equations and boundary conditions of the beam. Accuracy of the theory was verified by comparison to the results obtained with the help of other refined shear deformation theories. THAI and VO [10] dealt with bending, buckling, and vibration of nanobeams using a nonlocal sinusoidal shear deformation beam theory, thus avoiding the need of the use of the shear correction factors. The equations of motion and boundary conditions of the beam were derived with the use of the Hamilton's principle. The results obtained this way confirm the accuracy of the approach. AKGÖZ and CIVALEK [11] developed a new higher-order shear deformation model of a beam using the modified strain gradient theory, without the need of any shear correction factors. The authors studied static bending and free vibration of simply supported microbeams. The effects of thickness-tomaterial length ratio, slenderness ratio and shear deformation on deflections and natural frequencies of microbeams were analyzed. It was shown that the effect of shear deformation becomes more significant for less slender beams. SAWANT and DAHAKE [12] worked out a new hyperbolic shear deformation theory allowing for analysis of a thick beam bending. The theory enables calculating of the transverse shear stresses directly from the constitutive relations, using no shear correction factors. The present theory is verified on the example of a cantilever isotropic beam subjected to varying load. The results were compared with those based on the other theories. BOURADA et al. [13] considered bending and vibration of functionally graded beams. The authors developed a simple and refined trigonometric higherorder beam theory for this purpose, distinguished by fewer number of unknowns and equations than the other theories. The equations of motion were derived from the Hamilton's principle and the solutions were compared with those obtained based on other theories, demonstrating the effect of the inclusion of transverse normal strain on the deflections and stresses. POLIZZOTTO [14] considered a series of shear deformable beams using various approaches - from the Euler-Bernoulli to Timoshenko theory. Three deformation variables, namely two curvatures and a shear angle were applied with a view to determine two bending moments and the shear force. The principle of virtual power allowed to derive the equilibrium equations and the boundary conditions. In order to compute the normal stress the classical Navier expression was used, while the shear stress was calculated with the use of the Zhuravsky formula. The results are illustrated on the example of a simple cantilever beam. ENDO [15] presented a historical survey on the bending and shearing deformation concepts of beams and plates, compared with the corrected classical models. The survey shows that according to the corrected classical models the bending and shearing deflections are considered as separate physical entities. The method developed by the author enabled to carry out frequency analysis of isotropic plates and carbon fiber reinforced plastic laminated composite beams. The deformation concept enabled to develop a practical finite element formulation that is free from shear locking. ENDO [16] used the Hamilton's principle to derive the author's alternative beam and plate theories. The author assumed that the total deflection is a sum of the bending and shearing deflections. It was shown that according to these theories only a half of the shear deformations can be expressed explicitly. Such an approach is efficient in case of moderately thick structural models supplemented with the total deflection concept. ENDO [17] formulated exact frequency relationships between the classical Kirchhoff plate and the "one-half order" shear deformation plate or beam theory of the Mindlin plate theory in case of simply supported plate edges or beam ends. Additionally, an approximate frequency relationship was determined, having a very simple form and nearly enough accuracy for purposes of practical use. KHARLAB [18] proposed a complement to the theory of flexural shear stresses, being a generalization of the Zhuravsky's theory. The improved approach takes into account the deplanation of the beam cross section and allows to develop a simple formula for the potential energy of its deformation. GENOVESE and ELISHAKOFF [19] presented discussion of the role of the principle of virtual work and of the energy balance for formulation of planar static rod theories, in the range of large deformations, with consideration of the transverse shear. The discussion also included the differences between the Haringx and Engesser approaches to the problem. MAGNUCKI *et al.* [20] considered the simply supported beams subjected to non-uniformly distributed load. The model of the beams was formulated based on nonlinear hypothesis of deformation of the beam planar cross section and with consideration of the shear effect. Numerical-FEM models of the beams were developed with a view to compare the analytical and numerical results for an exemplary beam family. MAGNUCKI *et al.* [21] analytically studied the bending problem of the beams with consideration of a seventh-order shear deformation theory.

The shear effect in the beams, described in the publications mentioned above, is taken into account with the use of the first-, second- and third-order beam theory. Thus, the main purpose of this paper is presentation of an individual theory of planar cross section deformations of the beams formulated based on the Zhuravsky shear stress formula. The performance of this theory is demonstrated for bending of the example beams with various bisymmetrical cross sections. The paper is a continuation of the subject area presented by Magnucki *et al.* [20] and [21].

## 2. Bisymmetrical cross section of the beam – Zhuravsky shear stress

The object of the study is the beam with bisymmetrical cross section of depth h and length L (Fig.1). The beam is situated in the Cartesian coordinate system xyz.



Fig.1. Scheme of the bisymmetrical cross section of the beam

The symmetrically varying width of the cross section in the depth direction is assumed in the following form

$$b(y) = b \cdot \tilde{b}(\eta), \tag{1}$$

where the dimensionless width

$$\tilde{b}(\eta) = \beta_0 + (1 - \beta_0) (8\eta^2 - 16\eta^4)^{k_c},$$
(2)

and  $\beta_0 = b_0/b$  – parameter,  $k_c$  – exponent-real number,  $\eta = y/h$  – dimensionless coordinate. The Zhuravsky shear stress formula (Gere and Timoshenko [1]) is as follows

$$\tau_{xy}^{(Zh)}(x, y) = \frac{V(x)}{J_z} \frac{Q_z(y)}{b(y)},$$
(3)

where: V(x) – shear force, the second moment of the cross section about the z axis

$$J_{z} = \int_{-h/2}^{h/2} b(y) y^{2} dy = bh^{3} \widetilde{J}_{z}, \qquad (4)$$

the dimensionless second moment

$$\widetilde{J}_{z} = \int_{-1/2}^{1/2} \widetilde{b}(\eta) \eta^{2} d\eta , \qquad (5)$$

the first moment of the part of the cross section about the z axis

$$Q_{z}(y) = -\int_{-h/2}^{y} b(y_{1})y_{1}dy_{1} = bh^{2}\tilde{Q}_{z}(\eta),$$
(6)

the dimensionless first moment

$$\widetilde{Q}_{z}(\eta) = -\int_{-1/2}^{\eta} \widetilde{b}(\eta_{1})\eta_{1}d\eta_{1}, \qquad (7)$$

and  $\eta_1 = y_1/h$  – dimensionless coordinate  $(-1/2 \le \eta_1 \le \eta)$ , b < h – condition. Thus, the Zhuravsky shear stress formula (3) takes the following form

$$\tau_{xy}^{(Zh)}(x,\eta) = \frac{V(x)}{bh\tilde{J}_z} \frac{\tilde{Q}_z(\eta)}{\tilde{b}(\eta)}.$$
(8)

The formula (8) is a basis of formulation of the individual theory of planar cross section deformations of the beams.

#### 3. Analytical formulation of shear deformation theory of bending beams

A planar cross section before bending does not remain planar after bending of the beam (Fig.2).



Fig.2. Scheme of the deformation of the planar cross sections of the beam

The longitudinal displacement in accordance with Fig.2 is as follows

$$u(x,\eta) = -h \left[ \eta \frac{dv}{dx} - f_d(\eta) \psi(x) \right], \tag{9}$$

where:  $\psi(x) = u_1(x)/h$  – dimensionless displacement function,  $f_d(\eta)$  – deformation function of the planar cross section of the beam, v(x) – deflection.

Therefore, the longitudinal and shear strains are in the following form:

$$\varepsilon_{x}(x,\eta) = \frac{\partial u}{\partial x} = -h \left[ \eta \frac{d^{2}v}{dx^{2}} - f_{d}(\eta) \frac{d\psi}{dx} \right], \quad \gamma_{xy}(x,\eta) = \frac{dv}{dx} + \frac{\partial u}{\partial y} = \frac{df_{d}}{d\eta} \psi(x). \tag{10}$$

Consequently, the normal and shear stresses in accordance with Hooke's law take the following form:

$$\sigma_x(x,\eta) = -Eh\left[\eta \frac{d^2 v}{dx^2} - f_d(\eta) \frac{d\psi}{dx}\right], \qquad \tau_{xy}^{(An)}(x,\eta) = \frac{E}{2(1+\nu)} \frac{df_d}{d\eta} \psi(x), \tag{11}$$

where: E – Young's modulus,  $\nu$  – Poisson's ratio.

Equating the Zhuravsky shear stress formula (8) to the above shear stress expression (11b)  $\tau_{xy}^{(Zh)}(x,\eta) = \tau_{xy}^{(An)}(x,\eta)$ , after simple transformations one obtains:

• the derivative and deformation function

$$\frac{df_{d}}{d\eta} = \frac{1}{C_{0}} \frac{(1-4\eta^{2})\beta_{0} - 8(1-\beta_{0})J_{1}(\eta)}{\widetilde{b}(\eta)}, \ f_{d}(\eta) = \frac{1}{C_{0}} \int \frac{(1-4\eta^{2})\beta_{0} - 8(1-\beta_{0})J_{1}(\eta)}{\widetilde{b}(\eta)} d\eta \ (12)$$
  
where:  $J_{1}(\eta) = \int_{-1/2}^{\eta} (8\eta_{1}^{2} - 16\eta_{1}^{4})^{k_{c}} \eta_{1} d\eta_{1}, \ \text{and} \ C_{0} = \int_{0}^{1/2} \frac{(1-4\eta^{2})\beta_{0} - 8(1-\beta_{0})J_{1}(\eta)}{\widetilde{b}(\eta)} d\eta,$ 

• the dimensionless displacement function

$$\psi(x) = \frac{1}{4} (1+\nu) \frac{C_0}{\tilde{J}_z} \frac{V(x)}{Ebh}.$$
(13)

The derivative and deformation function (12) of the planar cross section of the beam satisfy the conditions:  $df_d/d\eta|_{\mp 1/2} = 0$  and  $f_d(\mp 1/2) = \mp 1$ .

The example cross sections of the beams (**CS-1**:  $\beta_0=2/10$ ,  $k_c=2$ , **CS-2** (**I-100**):  $\beta_0=4.5/50$ ,  $k_c=11.155$ , **CS-3** (**I-200**):  $\beta_0=7.5/90$ ,  $k_c=16.397$ ) and corresponding graphs of the deformation function (12b) are shown in Fig.3.



Fig.3. Three example cross sections and corresponding graphs of the function (12b)

In the particular case for rectangular cross section ( $\beta_0 = 1$ ,  $\tilde{b}(\eta) = 1$ ,  $\tilde{J}_z = 1/12$ ,  $C_0 = 1/3$ ), the derivative and the deformation function (12) of the planar cross section are as follows:

$$\frac{df_d}{d\eta} = 3\left(1 - 4\eta^2\right), \quad f_d(\eta) = \left(3 - 4\eta^2\right)\eta. \tag{14}$$

These above expressions (14) are consistent with the shear deformation theory presented for example by WANG *et al.* [4].

#### 4. Deflection of the beam with consideration of the shear deformation theory

The bending moment in accordance with the definition is as follows

$$M_{b}(x) = \int_{-h/2}^{h/2} y \sigma_{x}(x, y) b(y) dy .$$
(15)

Substituting the expressions for the dimensionless width (2) and normal stress (11a) into the above expression (15) and integrating, one obtains the equation

$$\widetilde{J}_{z}\frac{d^{2}v}{dx^{2}} - C_{v\psi}\frac{d\psi}{dx} = -\frac{M_{b}(x)}{Ebh^{3}},$$
(16)

where:  $C_{v\psi} = \int_{-1/2}^{1/2} f_d(\eta) \widetilde{p}(\eta) \eta d\eta$  – dimensionless coefficient.

Taking into account the paper MAGNUCKI *et al.* [20] the generalized load of the simply supported beam is assumed (Fig.4). The intensity of non-uniformly distributed load is in the following form

$$q(\xi) = \frac{k}{2\tanh(k/2)} \frac{1}{\cosh^2[k(0.5 - \xi)]} \frac{F}{L},$$
(17)

where:  $k - \text{parameter } k \in (0,\infty)$ ,  $\xi = x/L$  - dimensionless coordinate  $\xi \in (0,1)$  and F - total load of the beam, L - length of the beam.



Fig.4. Scheme of the simply supported beam under the generalized load

Consequently, the shear force is as follows

$$V(\xi) = \frac{1}{2\tanh(k/2)} \tanh[k(0.5 - \xi)]F, \qquad (18)$$

and the bending moment

$$M_{b}(\xi) = \frac{1}{2k \tanh(k/2)} \ln \frac{\cosh(k/2)}{\cosh[k(0.5 - \xi)]} FL.$$
(19)

The above expressions enable to formulate two particular cases of the load: the first one – *the uniformly distributed load* for  $k \rightarrow 0$ , and the second one – *three-point bending* for  $k \rightarrow \infty$  (Fig.5).



Fig.5. Graphs of the dimensionless shear forces  $\tilde{V}(\xi) = V(\xi)/F$  and dimensionless bending moments  $\tilde{M}_b(\xi) = M_b(\xi)/FL$ 

The differential equation (16) in dimensionless coordinate with consideration of the expression (19) takes the following form

$$\widetilde{J}_{z} \frac{d^{2}v}{d\xi^{2}} = C_{vv}L \frac{d\psi}{d\xi} - \frac{\lambda^{3}}{2k \tanh(k/2)} \ln \frac{\cosh(k/2)}{\cosh[k(0.5 - \xi)]} \frac{F}{Eb}, \qquad (20)$$

where:  $\lambda = L/h$  – relative length of the beam.

Integrating the equation (20) and taking into account the functions (13) and the condition  $dv/d\xi|_{1/2} = 0$ , one obtains the equation

$$\frac{dv}{d\xi} = \frac{\lambda^3}{2k \tanh(k/2)} \Phi_1(\xi) \frac{1}{\tilde{J}_z} \frac{F}{Eb},$$
(21)

where:

$$\Phi_{1}(\xi) = \frac{1}{4} (1+\nu) \frac{C_{0}C_{\nu\psi}}{\widetilde{J}_{z}\lambda^{2}} k \tanh[k(0.5-\xi)] + J_{\nu 1} - \int \ln \frac{\cosh(k/2)}{\cosh[k(0.5-\xi)]} d\xi, \qquad (22)$$

and  $J_{v1} = \int_{0}^{1/2} \ln \frac{\cosh(k/2)}{\cosh[k(0.5-\xi)]} d\xi$ .

Consequently, integrating the equation (21) and taking into account the condition v(0) = 0, one obtains the deflection curve of the beam

$$v(\xi) = \frac{\lambda^3}{2k \tanh(k/2)} \Phi_2(\xi) \frac{1}{\tilde{J}_z} \frac{F}{Ebh},$$
(23)

where:

$$\Phi_{2}(\xi) = \frac{1}{4} (1+\nu) \frac{C_{0}C_{\nu\psi}}{\tilde{J}_{z}\lambda^{2}} \ln \frac{\cosh(k/2)}{\cosh[k(0.5-\xi)]} + J_{\nu 1}\xi - \int \int \ln \frac{\cosh(k/2)}{\cosh[k(0.5-\xi)]} d\xi^{2} .$$
(24)

Thus, the maximum deflection  $\left(v_{\max}^{(An)} = v(1/2)\right)$  of the beam is in the following form

$$v_{\max}^{(An)} = \widetilde{v}_{\max}^{(An)} \frac{F}{Eb}, \qquad (25)$$

where: the dimensionless maximum deflection

$$\widetilde{v}_{\max}^{(An)} = \left(1 + \frac{C_s}{\lambda^2}\right) C_v \frac{\lambda^3}{\widetilde{J}_z},$$
(26)

the shear and bending coefficients

$$C_{s} = \frac{1}{2} (1+\nu) \frac{C_{0} C_{\nu\psi}}{\tilde{J}_{z}} \frac{\ln \left[\cosh(k/2)\right]}{J_{\nu 1} - 2J_{\nu 2}} \qquad C_{\nu} = \frac{J_{\nu 1} - 2J_{\nu 2}}{4k \tanh(k/2)}$$
(27)

and  $J_{v2} = \int_{0}^{1/2} \int \ln \frac{\cosh(k/2)}{\cosh[k(0.5-\xi)]} d\xi^2$ .

The expressions (26) and (27) make a basis of example studies of bending of the beams with consideration of the shear effect.

## 5. Analytical studies of beams bending

The exemplary analytical studies of beams bending are carried out for the beams of relative length  $\lambda$ =10 with three selected bisymmetrical cross sections (Fig.3): **CS-1**:  $\beta_0$ =2/10,  $k_c$ =2, **CS-2** (**I-100**):  $\beta_0$ =4.5/50,  $k_c$ =11.155, **CS-3** (**I-200**):  $\beta_0$ =7.5/90,  $k_c$ =16.397), and material constant – Poisson's ratio  $\nu$ =0.3. The dimensionless second moment and coefficients of these selected cross sections are specified in the Tab.1.

Cross section	CS-1	CS-2 (I-100)	CS-3 (I-200)
$\widetilde{J}_z$	0.060245	0.03420	0.02972
$C_0$	0.830938	1.290076	1.253780
$C_{v\psi}$	0.1457584	0.0762374	0.0653415

Table 1. The geometrical characteristic of selected cross section of beams

The studies are realized for two particular typical cases of the load (Fig.5):

• *the uniformly distributed load of intensity* q ( $k \rightarrow 0$ ), then,

the values of the shear and bending coefficients (27) are as follows

$$C_{s}^{(q)} = \frac{1}{2} (1+\nu) \frac{C_{0}C_{\nu\psi}}{\widetilde{J}_{z}} \lim_{k \to 0} \frac{\ln\left[\cosh(k/2)\right]}{J_{\nu 1} - 2J_{\nu 2}} = \frac{12}{5} (1+\nu) \frac{C_{0}C_{\nu\psi}}{\widetilde{J}_{z}}, \quad C_{\nu}^{(q)} = \lim_{k \to 0} \frac{J_{\nu 1} - 2J_{\nu 2}}{4k \tanh(k/2)} = \frac{5}{384}. \quad (28)$$

The values of the above shear coefficient and dimensionless maximum deflections (26) of the example beams are specified in Table 2.

Table 2. The calculation results of the selected beams under uniformly distributed load

Cross section	CS-1	CS-2 (I-100)	CS-3 (I-200)
$C_s^{(q)}$	6.272	8.972	8.600
${\cal V}_{ m max}^{(An-q)}$	229.69	414.88	475.80

• *three-point bending*  $(k \rightarrow \infty)$ , then,

the values of the shear and bending coefficients (27) are as follows

$$C_{s}^{(F)} = \frac{1}{2} (1+\nu) \frac{C_{0}C_{v\psi}}{\widetilde{J}_{z}} \lim_{k \to \infty} \frac{\ln[\cosh(k/2)]}{J_{v1} - 2J_{v2}} = 3(1+\nu) \frac{C_{0}C_{v\psi}}{\widetilde{J}_{z}}, \quad C_{v}^{(F)} = \lim_{k \to \infty} \frac{J_{v1} - 2J_{v2}}{4k \tanh(k/2)} = \frac{1}{48}.$$
(29)

The values of the above shear coefficient and dimensionless maximum deflections (26) of the example beams are specified in Table 3.

Table 3. The calculation results of the selected beams under three-point bending

Cross section	CS-1	CS-2 (I-100)	<b>CS-3 (I-200)</b>
$C_s^{(F)}$	7.840	11.216	10.750
$v_{\max}^{(An-F)}$	372.92	677.48	776.35

The dimensionless shear stress based on the expression (11b) for value  $\tilde{V}(\xi) = 1/2$  of the dimensionless shear force is as follows

$$\tilde{\tau}_{xy}^{(An)}(\eta) = \frac{1}{16\tilde{J}_z} \frac{(1-4\eta^2)\beta_0 - 8(1-\beta_0)J_1(\eta)}{\tilde{b}(\eta)}.$$
(30)

The graphs of the dimensionless shear stress (30) for the three selected bisymmetrical cross sections are shown in Fig.6.



Fig.6. Graphs of the dimensionless shear stresses in the selected bisymmetrical cross sections

## 6. Numerical-FEM studies of beams bending

The considered beams are modelled using the SolidWorks software package. Taking into account the symmetry of the beams and their loads only a quarter of the whole structure is analyzed (Fig. 7a). The boundary conditions imposed on the model guarantee its proper behaviour. The model of the stockiest of the beams (i.e. CS1) is composed of about 572 thousand of 3D tetrahedral finite elements with 4 Jacobian points. Such an approach corresponds to the simplest integrating at the element area, being sufficient in case of the fine mesh applied in the computation.

The number of the nodes reaches nearly 840 000. In case of the CS2 and CS3 variants these numbers are correspondingly smaller. A part of the CS1 mesh is shown in Fig. 7b. The CS1 mesh could be coarser without harming the computation quality. Nevertheless, in order to maintain equal mesh quality in all the considered cases, the mesh scale does not vary.

The origin of the beam Cartesian coordinate system is located in the beginning of the beam neutral axis. The *x*-axis coincides with the neutral axis, the *y*-axis points down and, in consequence, *z*-axis is normal to the longitudinal middle plane of the beam.

The boundary conditions specified below ensure that the one fourth of the beam behaves the same as if it belonged to the whole beam:

- the beam model is simply supported at its edge (i.e. for *x*=0), therefore, the *y* displacements of the wall lying in the *yz*-plane are equal to zero,
- the *x* displacements of the middle wall of the beam, i.e. of the middle surface  $S_m$  of the model (being the transverse plane of symmetry of the beam) are equal to zero too,
- the *z* displacements of the model lying in the *xy*-plane (for *z*=0, the longitudinal plane of symmetry of the beam) are zero.



Fig.7. a) – The FEM-model of the beam, b) – An example its mesh (confined approximately to the part marked with the dotted circle in 7a)

Two load cases are considered:

- 1. three-point bending, with the force  ${}^{1}\!\!\!/_{4}F$  downward directed and applied to the  $S_{m}$  surface;

Results of the numerical study are specified in Table 4.

Table 4. The results of the numerical-FEM calculation of the selected beams

Cross section	CS-1	CS-2 (I-100)	<b>CS-3 (I-200)</b>
$v_{\max}^{(FEM-q)}$	229.6	414.9	475.4
$v_{\max}^{(FEM-F)}$	372.8	676.6	775.2

Comparison of the results obtained with FEM (Table 4) with data of the Tables 2 and 3 displays very good compliance of the analytical and numerical approach to the problem. The differences do not exceed 0.5%.

The graphs and values of the dimensionless shear stresses obtained with FEM are identical with those shown in Fig 6.

# 7. Conclusions

The theory of planar cross section deformations of the beams adopted based on the Zhuravsky shear stress formula well depicts the shear effect in the bent beams. The differences between the maximum deflections and shear stresses calculated analytically and numerically are insignificant (below 0.5%).

The proposed theory is an original idea and makes a generalization of earlier known beam theories. Therefore, it may be easily used in analytical modeling of the beams with consideration of the shear effect. It should be noted that the higher-order theories introduce additional unknowns that give rise to difficulties in the analytical research, as indicated by Wang *et al.* [4].

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